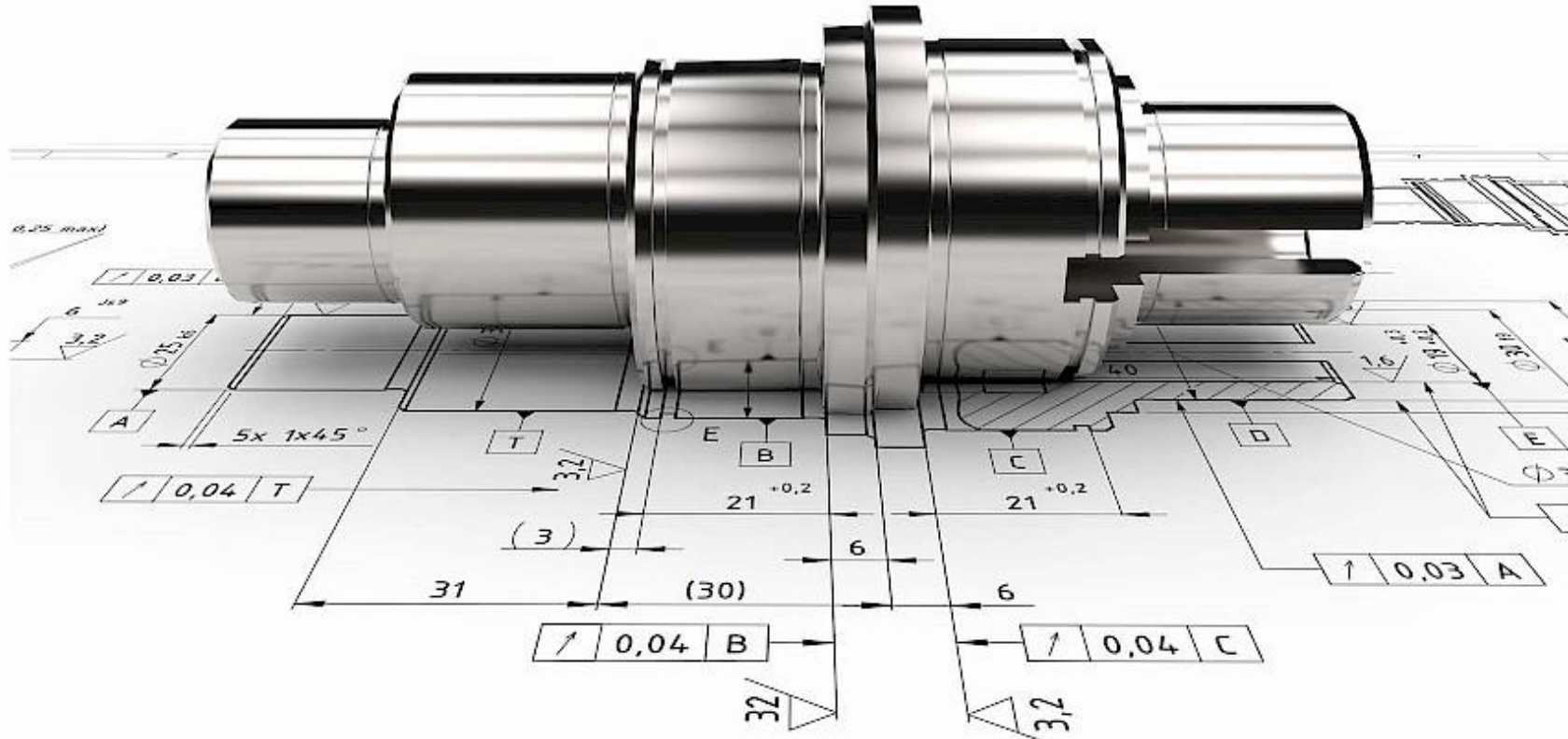
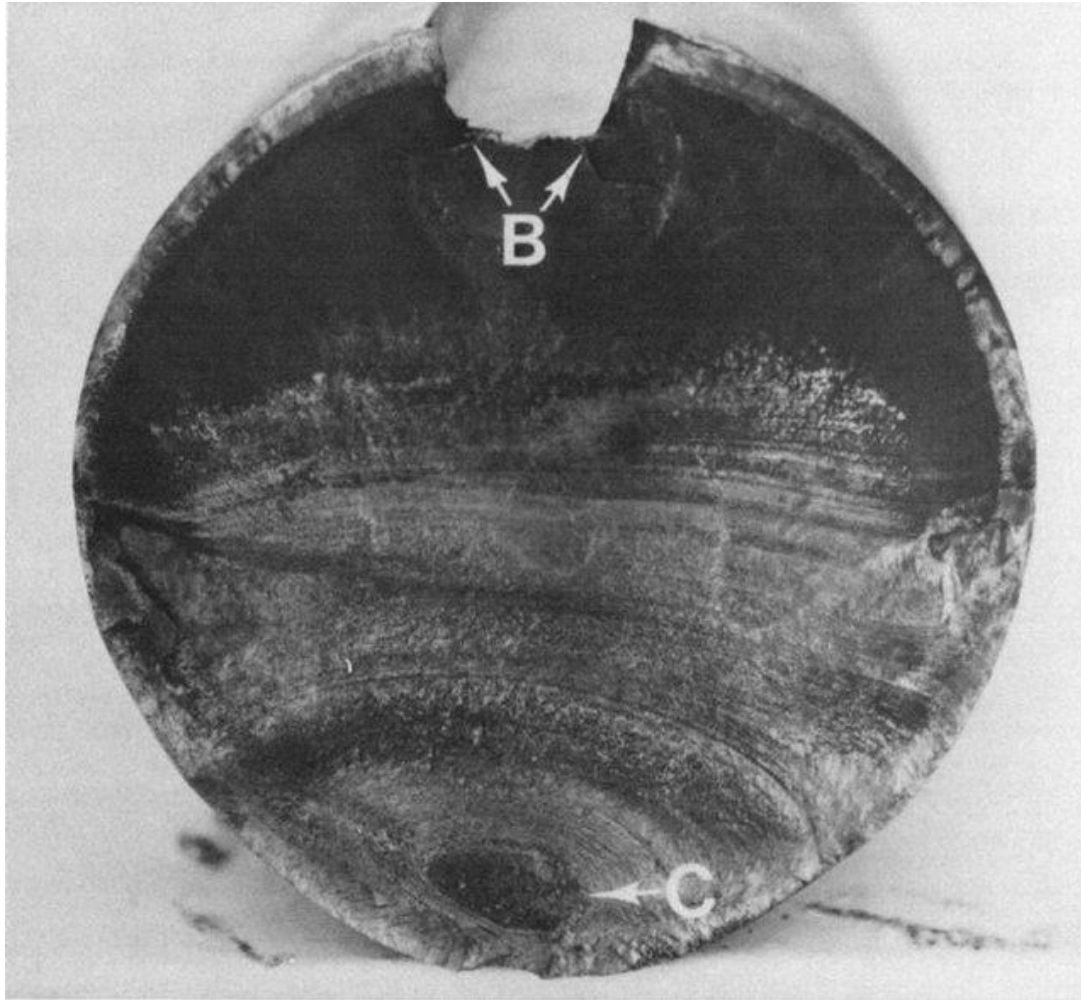


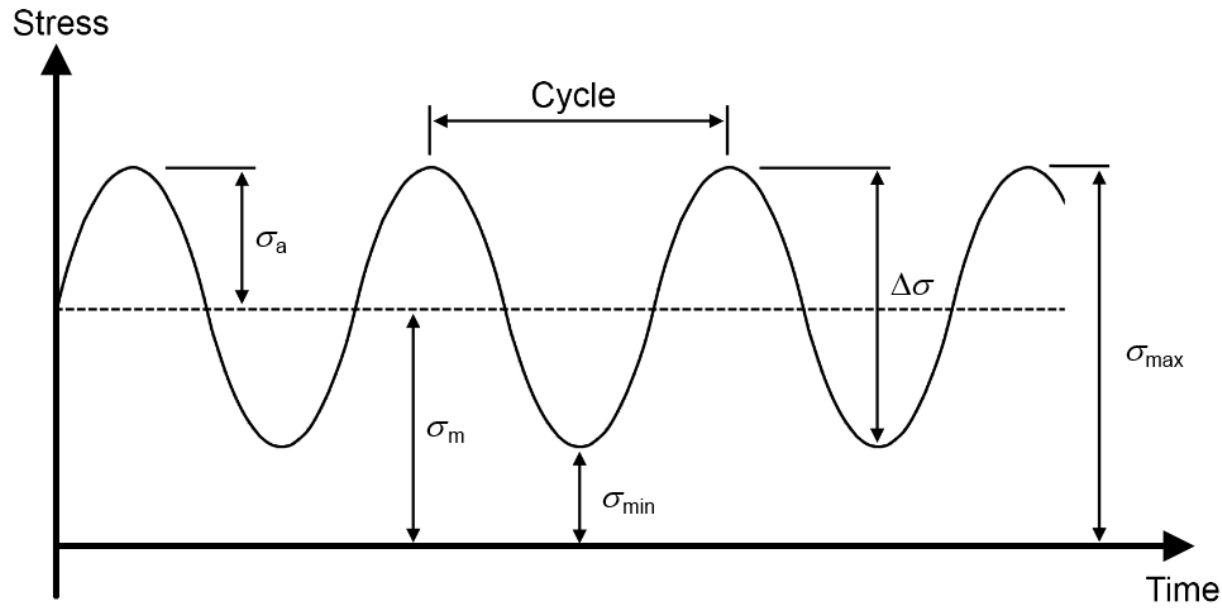
# Shafts, axels\_3



# FATIGUE STRENGTH



The failure began at the end of a keyway that was machined without fillets (B) and progressed to final rupture at (C). The final rupture zone is small, indicating that loads were low.



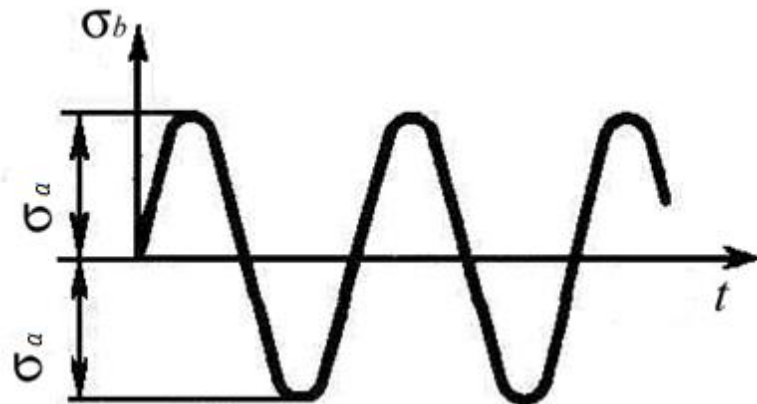
$$\sigma = \sigma_c + \sigma_a \sin \omega t$$

$$\sigma_m = \frac{\sigma_{\max} + \sigma_{\min}}{2} \quad \text{Mean stress}$$

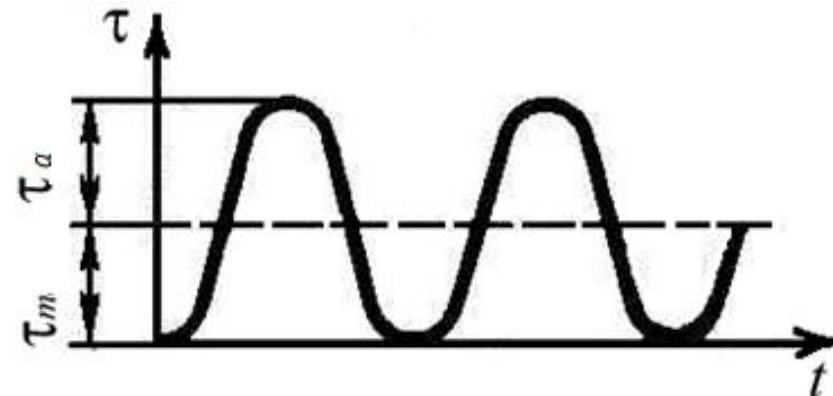
$$\sigma_a = \frac{\sigma_{\max} - \sigma_{\min}}{2} \quad \text{Stress Amplitude (Alternating stress)}$$

$$r = \frac{\sigma_{\min}}{\sigma_{\max}} \quad \text{Stress ratio}$$

***Changing of bending stresses***



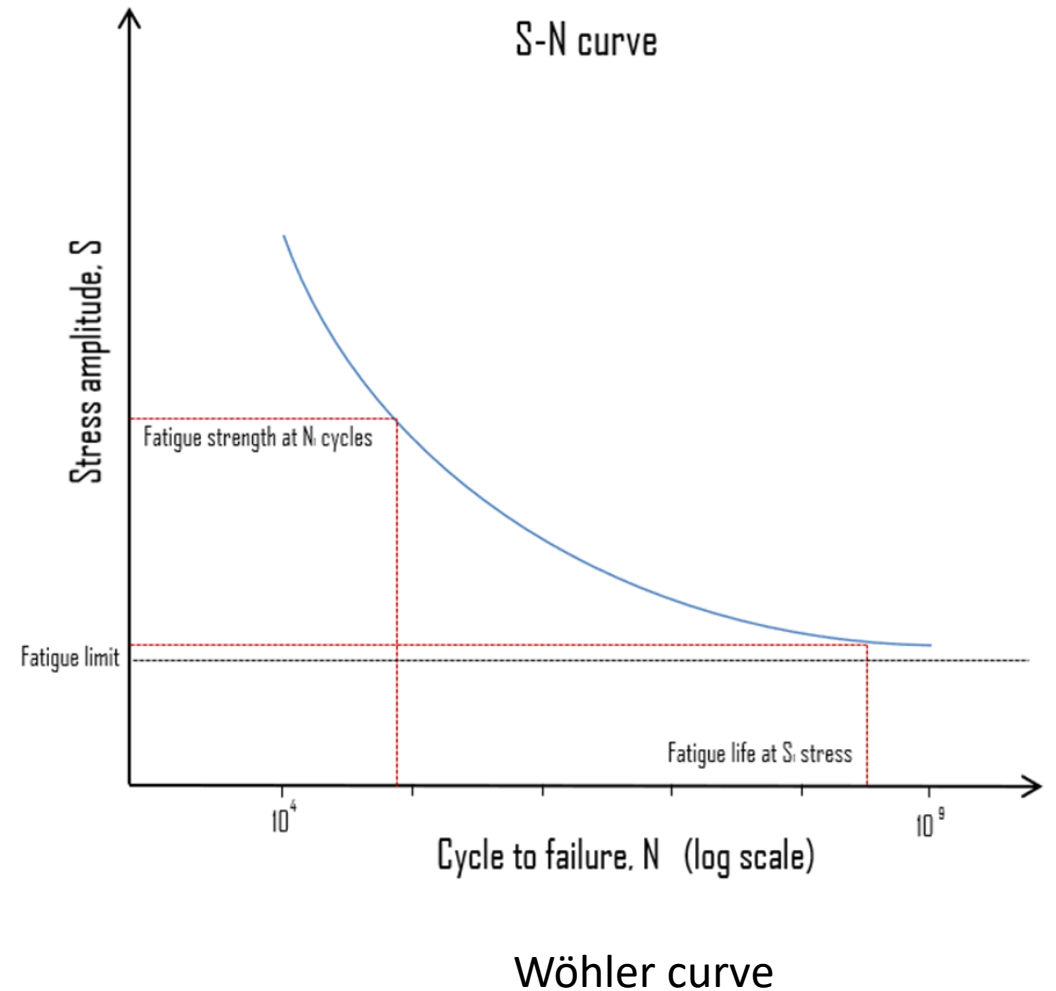
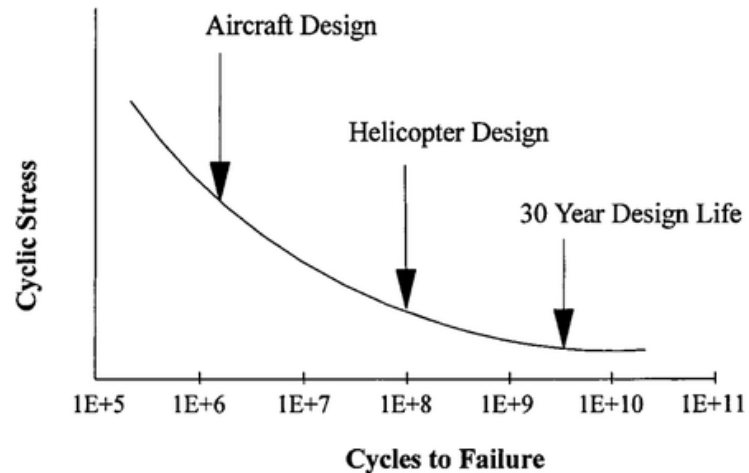
***Changing of torsion stresses***



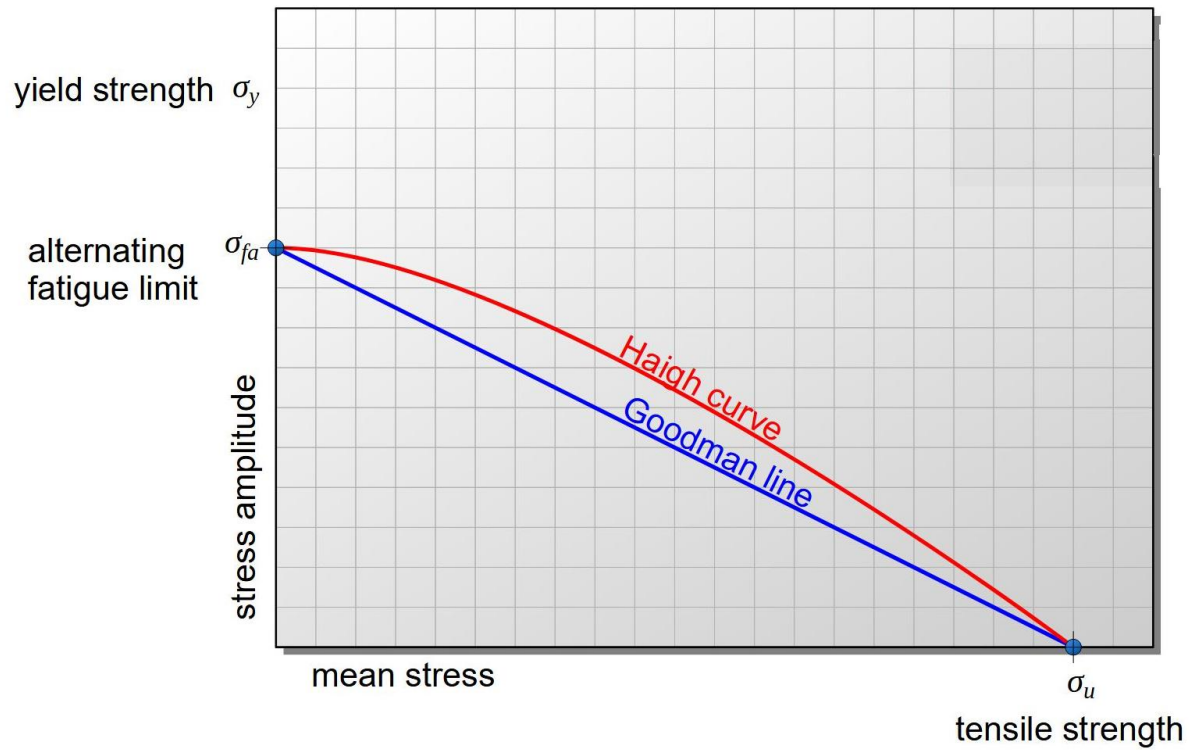
**Fatigue Limit (endurance limit)** is the stress level, below which fatigue failure does not occur. This limit exists only for some ferrous (iron-base) and titanium alloys, for which the S–N curve becomes horizontal at higher N values. Other structural metals, such as aluminium and copper, do not have a distinct limit and will eventually fail even from small stress amplitudes.

**Fatigue Strength** - the value of stress at which failure occurs after some specified number of cycles

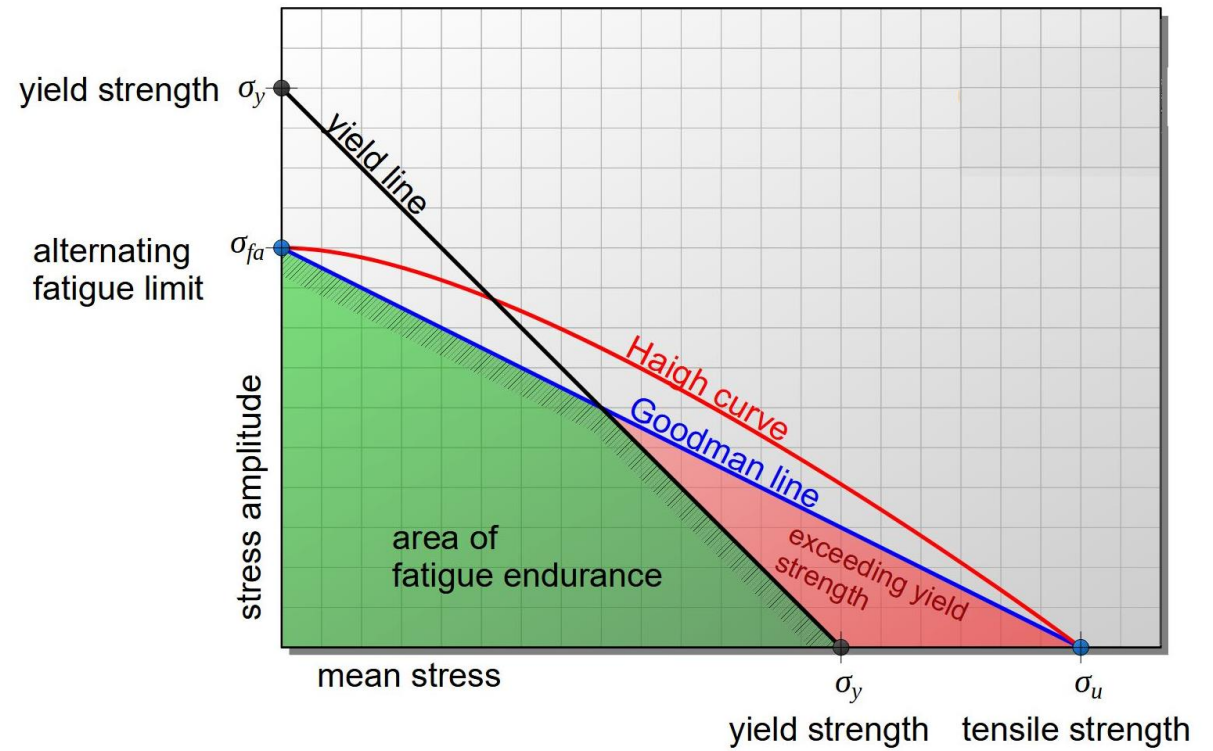
**Fatigue Life** characterizes a material's fatigue behavior. It is the number of cycles to cause failure at a specified stress level



## Haigh Diagram (Goodman diagram)



Goodman line 
$$\frac{\sigma_a}{\sigma_{fa}} + \frac{\sigma_m}{\sigma_u} = 1$$



Yield line 
$$\frac{\sigma_a}{\sigma_y} + \frac{\sigma_m}{\sigma_y} = 1$$

$$S \geq [S] \quad \text{Safety factor}$$

$$S_{\sigma} = \frac{\sigma_{fa}}{K_{\sigma d} \cdot \sigma_{\max}} \quad S_{\tau} = \frac{\tau_{fa}}{K_{\tau d} \cdot \tau_{\max}}$$

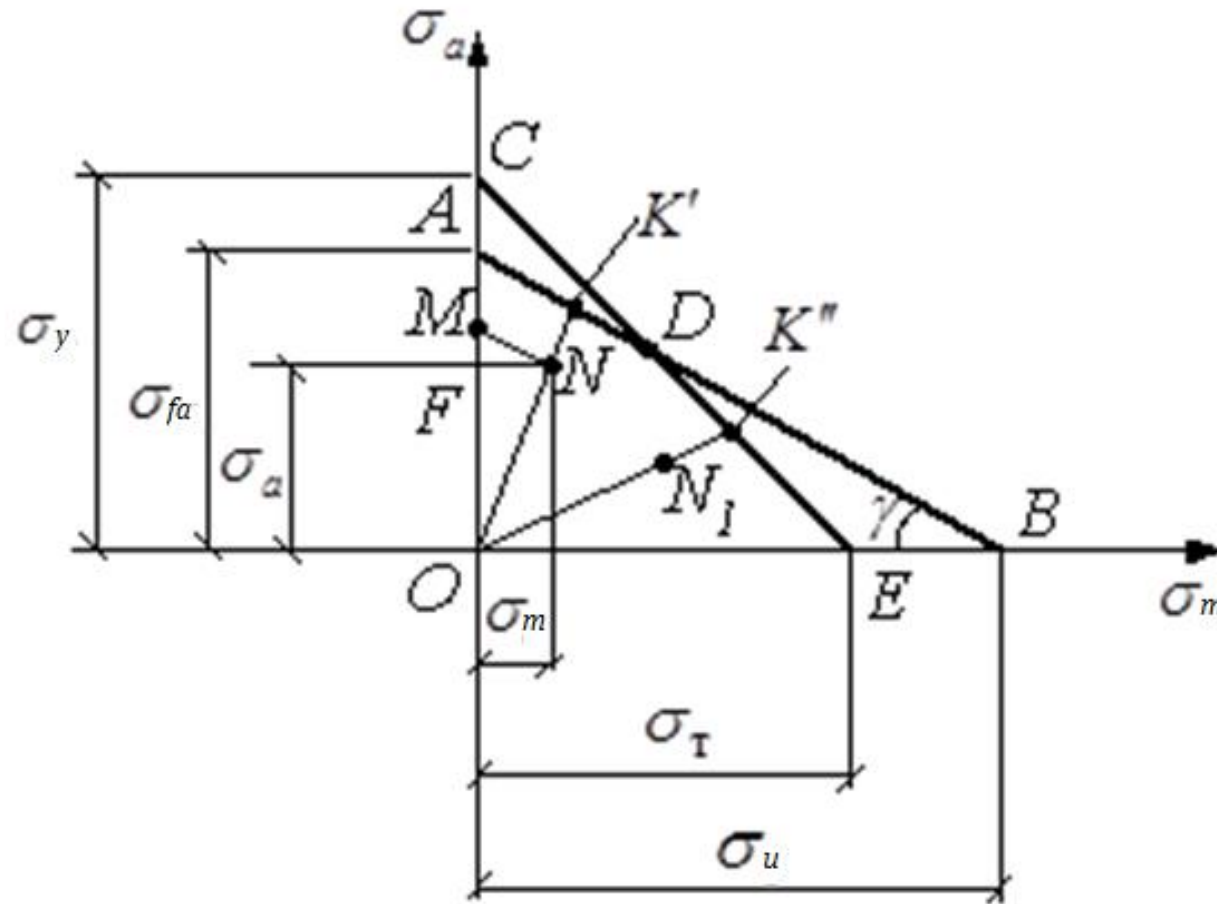
$$S_{\sigma} = \frac{\sigma'_{\max}}{\sigma_{\max}} = \frac{\sigma'_m + \sigma'_a}{\sigma_m + \sigma_a} = \frac{OK'}{ON}$$

$$\frac{OK'}{ON} = \frac{OA}{OM}$$

$$OA = \sigma_{fa}$$

$$OM = OF + FM = \sigma_a + \sigma_m \cdot \operatorname{tg} \gamma$$

$$\psi_{\sigma} = \operatorname{tg} \gamma = \frac{\sigma_{fa}}{\sigma_u}$$



$$S_{\sigma} = \frac{\sigma_{fa}}{K_{\sigma d} \sigma_a + \psi_{\sigma} \sigma_m}$$

$$K_{\sigma d} = \frac{K_{\sigma}}{K_b \cdot K_a}$$

$$K_{\tau d} = \frac{K_{\tau}}{K_b \cdot K_a}$$

the total factors of fatigue strength reducing

$K_{\sigma}, K_{\tau}$  stress concentration factors

$K_a$  surface roughness factor

$K_b$  scale factor

Obtained experimentally, analytically, etc

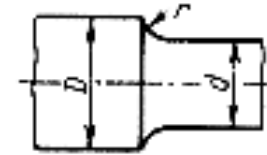
Published in charts and tables

<i>Diameter (d) (mm)</i>	<i>K<sub>b</sub></i>
$d \leq 7.5$	1.00
$7.5 < d \leq 50$	0.85
$d > 50$	0.75

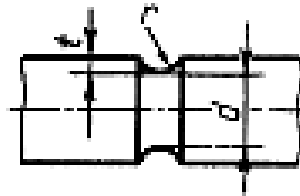
*Values of size factor*

## The most typical stress concentrations of the shaft

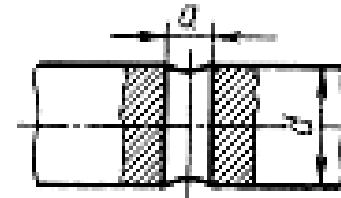
- Filleted transition regions;



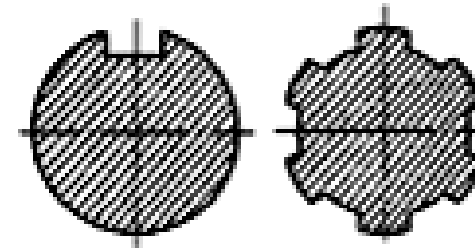
- Grooves;



- Radial holes;



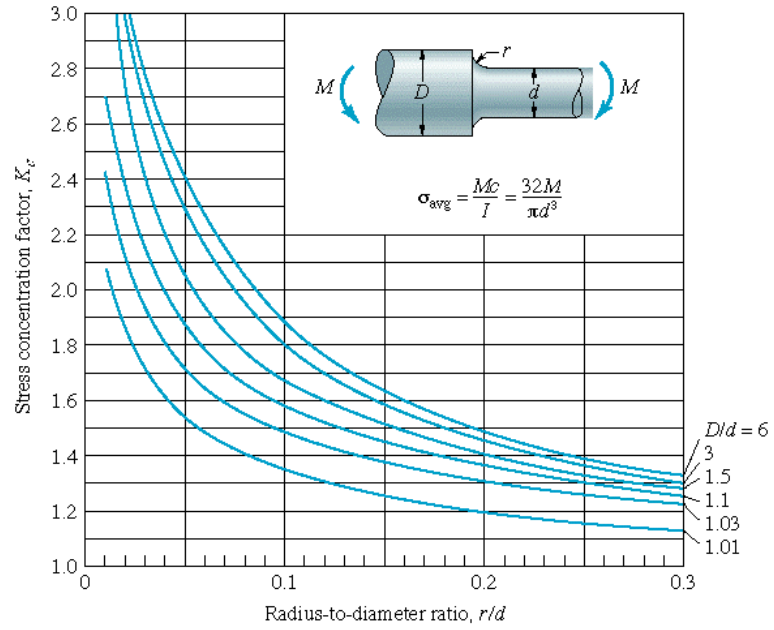
- Keyed and splined portions;



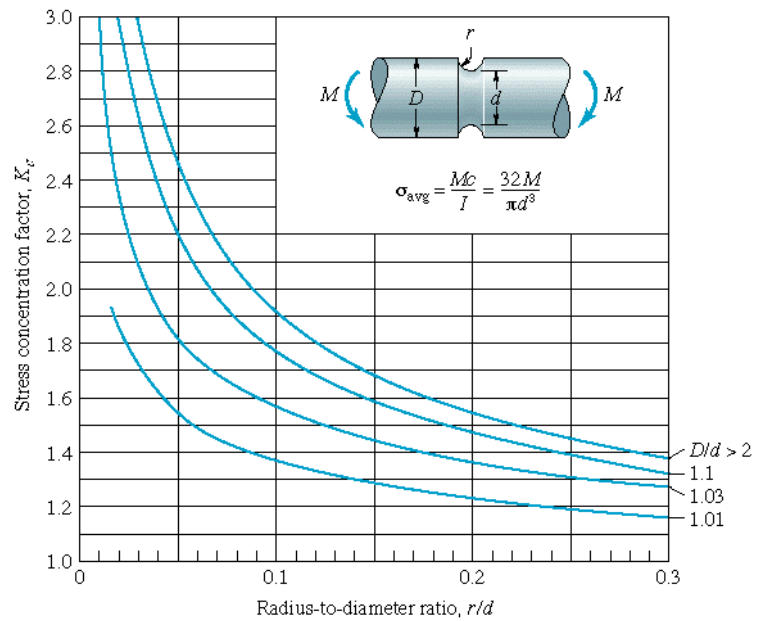
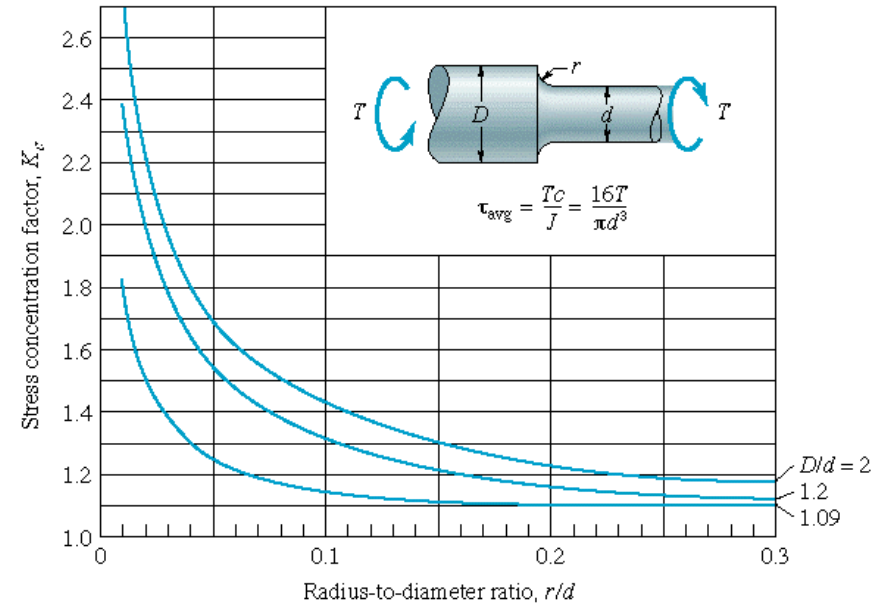
- Threaded portions;

- Interference fits.

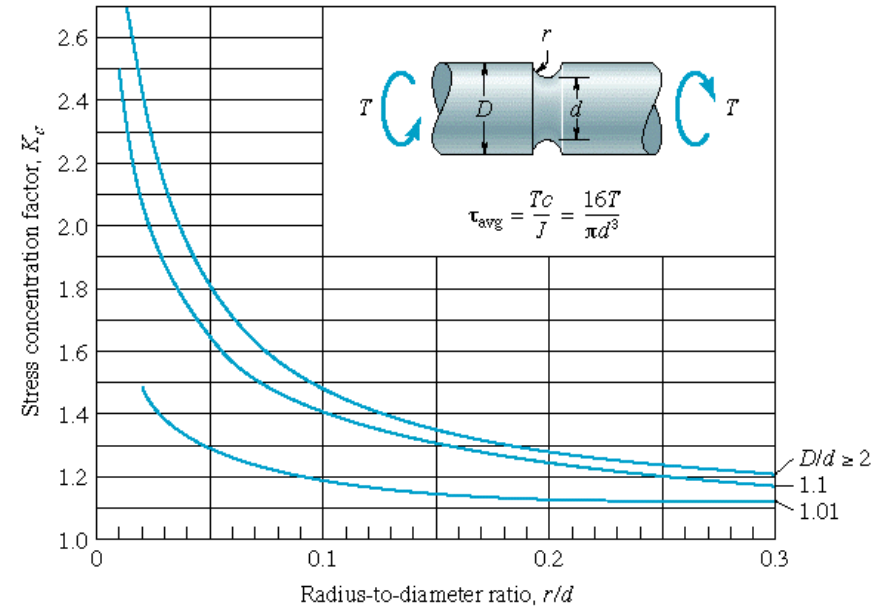




(b)



(b)



(c)

*Safety factor for bending*

$$S_{\sigma} = \frac{\sigma_{fa}}{\frac{K_{\sigma}}{K_b \cdot K_a} \cdot \sigma_a + \psi_{\sigma} \cdot \sigma_m}$$

*Safety factor for torsion*

$$S_{\tau} = \frac{\tau_{fa}}{\frac{K_{\tau}}{K_b \cdot K_a} \cdot \tau_a + \psi_{\tau} \cdot \tau_m}$$

*Safety factor*

$$S = \frac{S_{\sigma} \cdot S_{\tau}}{\sqrt{S_{\sigma}^2 + S_{\tau}^2}} \geq [S] = 1.5 \dots 2.5.$$

$$S_{\sigma} = \frac{\sigma_{fa}}{\frac{K_{\sigma}}{K_b \cdot K_a} \cdot \sigma_a + \psi_{\sigma} \cdot \sigma_m} \qquad S_{\tau} = \frac{\tau_{fa}}{\frac{K_{\tau}}{K_b \cdot K_a} \cdot \tau_a + \psi_{\tau} \cdot \tau_m}$$

$\sigma_{fa}, \tau_{fa}$  – limit of endurance in bending and in torsion (table-material)

$\sigma_{fa} = 0.43 \cdot \sigma_{ult}$  - for carbon steels;

$\sigma_{fa} = 0.35 \cdot \sigma_{ult} + 120$  - for alloy steels;

$$\tau_{fa} = (0.2 \dots 0.3) \cdot \sigma_{ult} \cdot$$

$\sigma_a, \tau_a$  – alternating bending and torsion stresses

$$\sigma_a = \frac{\sigma_{max} - \sigma_{min}}{2} = \sigma_{max} = \frac{M_{\Sigma}}{W} = \frac{M_{\Sigma}}{0.1 \cdot d^3};$$

$$\tau_a = \frac{\tau_{max} - \tau_{min}}{2} = \frac{\tau_{max}}{2} = 0.5 \cdot \frac{T}{W_p} = 0.5 \cdot \frac{T}{0.2 \cdot d^3} \cdot$$

$\sigma_m, \tau_m$  – mean bending and torsion stresses

$$\sigma_{mean} = \frac{\sigma_{max} + \sigma_{min}}{2} = 0;$$

$$\tau_{mean} = \frac{\tau_{max} + \tau_{min}}{2} = \frac{\tau_{max}}{2} = 0.5 \cdot \frac{T}{W_p} = 0.5 \cdot \frac{T}{0.2 \cdot d^3}.$$

$\psi_\sigma, \psi_\tau$  – factors of constant components of bending and torsion stresses (table-material)

$\psi_\sigma = 0.1; \psi_\tau = 0.05$  – for carbon steels;

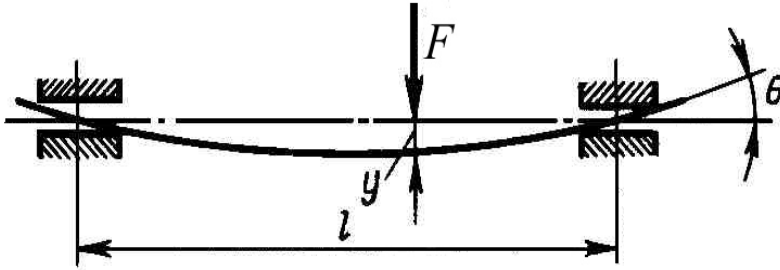
$\psi_\sigma = 0.15; \psi_\tau = 0.1$  – for alloy steels.

# STIFFNESS ANALYSIS OF THE SHAFT

## Flexural stiffness

Basic criteria of flexural stiffness are:

- Maximum deflection (sag)  $y$  of the shaft;
- Angle of rotation  $\theta$  of support sections.



*Flexural stiffness conditions*

$$y \leq [y]; \quad \theta \leq [\theta],$$

where

$[y]$  is the maximum safe sag;  $[\theta]$  is the maximum safe angle of rotation.

$[y] = 0.01m$  – for shafts of spur gears and worm gear drives;

$[y] = 0.005m$  – for shafts of bevel gear, hypoid gear and hourglass drives;

worm gear

$[y] = (0.0002 \dots 0.0003)l$  – for general purpose shafts used in machine tools;

$[\theta] = 0.001 \text{ rad}$  – for shafts mounted in sliding contact bearings;

$[\theta] = 0.005 \text{ rad}$  – for shafts mounted in radial ball bearings.

## Torsional stiffness

*Basic criterion of torsional rigidity is the angle of twist.*

### ***Torsional stiffness condition***

$$\varphi \leq [\varphi],$$

where  $[\varphi]$  is the maximum safe angle of twist.

$$\varphi = \frac{T \cdot l}{G \cdot J_p},$$

where  $T$  is torque;  $l$  is length of the shaft;  $G$  is shear modulus;  
 $J_p = \pi d^4/32$  is polar moment of inertia.

